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DEPARTMENT OF HOMELAND SECURITY

8 CFR Part 274a

[BCIS No. 2152–01]

RIN 1615–AA63

Employment Authorization Documents; Correction

AGENCY: Bureau of Citizenship and Immigration Services, DHS.

ACTION: Interim rule; correction.

SUMMARY: The Department of Homeland Security (DHS) published in the *Federal Register* of July 30, 2004, an interim rule which amended the DHS regulations governing issuance of Employment Authorization Documents (EADs). The interim rule contained an error that is corrected in this document.

DATES: This correction is effective July 30, 2004.

FOR FURTHER INFORMATION CONTACT: Jonathan Mills, Residence and Status Services, Office of Program and Regulations Development, Bureau of Citizenship and Immigration Services, Department of Homeland Security, 425 “I” Street, NW., ULLICO Building, Third Floor, Washington, DC 20536, telephone (202) 514–4754.

SUPPLEMENTARY INFORMATION:

Need for Correction

As published in the *Federal Register* on July 30, 2004 (69 FR 45555), the interim rule amending part 274a contains an error that is in need of correction.

Correction of Publication

■ Accordingly, the publication on July 30, 2004 (69 FR 45555), of the interim rule that was the subject of FR Doc. 04–16938 is corrected as follows:

PART 274a—CONTROL OF EMPLOYMENT OF ALIENS

§ 274a.12 [Corrected]

■ 1. On page 45557, in the first column, amendatory instruction 2d is corrected to read: “Revising paragraph (c), introductory text;”

Dated: August 3, 2004.

Richard A. Sloan,

Director, Regulations and Forms Services Division.

[FR Doc. 04–17971 Filed 8–5–04; 8:45 am]

BILLING CODE 4410–10–P

DEPARTMENT OF TRANSPORTATION (DOT)

Federal Aviation Administration

14 CFR Part 39

[Docket No. FAA–2004–18758; Directorate Identifier 2004–NE–24–AD; Amendment 39–13763; AD 2004–16–07]

RIN 2120–AA64

Airworthiness Directives; General Electric Company (GE); CT7–2D1 Turboshaft Engines.

AGENCY: Federal Aviation Administration (FAA), DOT.

ACTION: Final rule; request for comments.

SUMMARY: The FAA is adopting a new airworthiness directive (AD) for GE CT7–2D1 turboshaft engines. This AD requires replacing certain turbine stage 2 aft cooling plates, part number (P/N) 6064T0P02. This AD results from an uncontained failure of a turbine stage 2 aft cooling plate in a GE CT7 turboprop engine. We are issuing this AD to prevent a similar uncontained failure of turbine stage 2 aft cooling plates in GE CT7–2D1 turboshaft engines.

DATES: This AD becomes effective August 23, 2004.

We must receive any comments on this AD by October 5, 2004.

ADDRESSES: Use one of the following addresses to submit comments on this proposed AD.

- DOT Docket Web site: Go to <http://dms.dot.gov> and follow the instructions for sending your comments electronically.
- Government-wide rulemaking Web site: Go to <http://www.regulations.gov>

and follow the instructions for sending your comments electronically.

- Mail: Docket Management Facility, U.S. Department of Transportation, 400 Seventh Street, SW., Nassif Building, Room PL–401, Washington, DC 20590–0001.

- Fax: (202) 493–2251.

- Hand Delivery: Room PL–401 on the plaza level of the Nassif Building, 400 Seventh Street, SW., Washington, DC, between 9 a.m. and 5 p.m., Monday through Friday, except Federal holidays.

You may examine the comments on this AD in the AD docket on the Internet at <http://dms.dot.gov>.

FOR FURTHER INFORMATION CONTACT:

Mark J. Bouyer, Aerospace Engineer, Engine Certification Office, FAA, Engine & Propeller Directorate, 12 New England Executive Park, Burlington, MA 01803–5299; telephone (781) 238–7757; fax (781) 238–7755.

SUPPLEMENTARY INFORMATION: In July of 1999, an uncontained failure of a turbine stage 2 aft cooling plate occurred on a GE CT7 turboprop engine. An investigation indicated that the electro-discharge (EDM) machining of the cooling airholes caused microcracks in the walls of the airholes that could propagate through the turbine stage 2 cooling plate and result in an uncontained engine failure. We issued AD 2002–01–03 to prevent an uncontained failure of turbine stage 2 aft cooling plates in GE CT7 turboprop engines. In October of 2003, the manufacturer informed us of a similar problem with GE CT7–2D1 turboshaft engines. This AD requires replacing turbine stage 2 aft cooling plates, P/N 6064T07P02, with serial numbers (SNs) GFFN****, GFFP****, GFFR0*** through GFFR7***, GFFR81** through GFFR89**, GFFR8A** through GFFR8G**, GFFR8H92 through GFFR8H99, and GFFR8H9A through GFFR8H9N. Asterisks represent any subsequent number or letter that follow the root SN. This condition, if not corrected, could result in an uncontained failure of turbine stage 2 aft cooling plates in GE CT7–2D1 turboshaft engines.

FAA’s Determination and Requirements of This AD

Although no aircraft that are registered in the United States use these engines, the possibility exists that the engines could be used on aircraft that